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Comparison of Relative Motion Control Algorithms for Point Capturing of Space Debris Object

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Assumptions:

- Active spacecraft is in the vicinity of space debris
- The relative rotational and translational motion is assumed to be known
- The active spacecraft is equipped:
 - with thrusters to control translational motion and reaction wheels to control attitude motion
 - with a capturing system (robotic arm, magnetic capturing system etc.)
- Passive space debris object is tumbling
- It is possible to catch the debris at one specific point on the debris surface

Aims of the work:

- Develop different control algorithms for debris approaching
- Compare the performance of the algorithms





Equations of relative motion





Dynamical equations of relative rotational motion





Relative translational motion

Equations of relative translational motion:

$$\ddot{x}_{ij} - 2\omega_{OT} \dot{y}_{ij} - \dot{\omega}_{OT} y_{ij} - 3\omega_{OT}^2 x_{ij} = a_x + d_x$$

$$\ddot{y}_{ij} + 2\omega_{OT} \dot{x}_{ij} + \dot{\omega}_{OT} x_{ij} = a_y + d_y$$

$$\ddot{z}_{ij} + \omega_{OT}^2 z_{ij} = a_z + d_z$$

Relative position of the points:

 $\boldsymbol{\rho}_{ij} = \boldsymbol{\rho}_0 + \mathbf{r}_S^i - \mathbf{r}_T^j$ $\dot{\boldsymbol{\rho}}_{ij} = \dot{\boldsymbol{\rho}}_0 + \boldsymbol{\omega} \times \mathbf{r}_S^i$ $\ddot{\boldsymbol{\rho}}_{ij} = \ddot{\boldsymbol{\rho}}_0 + \dot{\boldsymbol{\omega}} \times \mathbf{r}_S^i + \boldsymbol{\omega} \times (\boldsymbol{\omega} \times \mathbf{r}_S^i)$



Desired relative position for capturing

• Relative directions

$$\mathbf{e}^{T} = \mathbf{D}_{S}^{T} \mathbf{e}_{S}^{S} + \mathbf{e}_{T}^{T}, \ \mathbf{e}_{S} = \frac{\mathbf{r}_{S}^{i}}{\left|\mathbf{r}_{S}^{i}\right|}, \ \mathbf{e}_{T} = \frac{\mathbf{r}_{T}^{j}}{\left|\mathbf{r}_{T}^{j}\right|}$$

- For capturing must be satisfied: $\mathbf{e} = 0$ \mathbf{H} $\dot{\mathbf{e}} = 0$
- Desired position of center of mass:

 $\boldsymbol{\rho}_{d}^{T} = \boldsymbol{K} \cdot \boldsymbol{e}_{T}^{T}, \quad \boldsymbol{K} > \left| \boldsymbol{r}_{C}^{i} \right| + \left| \boldsymbol{r}_{T}^{j} \right|$

• Dynamical equation:

$$\left(\frac{d^2\mathbf{e}}{dt^2}\right)^T = \left(\dot{\mathbf{\omega}}^{\times} + \mathbf{\omega}^{\times}\mathbf{\omega}^{\times}\right)\left(\mathbf{e}^T - \mathbf{e}_T^T\right)$$





SDRE-based control algorithm

Nonlinear dynamical system

 $\dot{\mathbf{x}} = \mathbf{f}(\mathbf{x}) + \mathbf{g}(\mathbf{x},\mathbf{u})$

• After linearization

$$\dot{\mathbf{x}} = \mathbf{A}(\mathbf{x})|_{\mathbf{X}(t_k)} \mathbf{x} + \mathbf{B}(\mathbf{x})|_{\mathbf{X}(t_k)} \mathbf{u}$$

• Riccati equation

$$\mathbf{P}(\mathbf{x})\mathbf{A}(\mathbf{x}) + \mathbf{A}^{T}(\mathbf{x})\mathbf{P}(\mathbf{x}) - \mathbf{P}(\mathbf{x})\mathbf{B}(\mathbf{x})\mathbf{R}^{-1}\mathbf{B}^{T}(\mathbf{x})\mathbf{P}(\mathbf{x}) + \mathbf{Q} = 0$$

• Minimization of the functional

$$J = \frac{1}{2} \int_{0}^{t_f} \left[\mathbf{x}(t)^T \mathbf{Q} \mathbf{x}(t) + \mathbf{u}(t)^T \mathbf{R} \mathbf{u}(t) \right] dt$$

• Control law

$$\mathbf{u}(\mathbf{x}) = -\mathbf{R}^{-1} \left(\mathbf{B}^{T}(\mathbf{x}) \mathbf{P}(\mathbf{x}) \right) \mathbf{x}$$



State-dependent coefficients for dynamical system

• For relative angular motion

$$\begin{bmatrix} \dot{\mathbf{e}} \\ \ddot{\mathbf{e}} \end{bmatrix}_{6\times 1} = \begin{bmatrix} 0 & \mathbf{E} \\ \left(\left(\dot{\boldsymbol{\omega}} \right)^{\times} + \boldsymbol{\omega}^{\times} \boldsymbol{\omega}^{\times} \right) & 0 \end{bmatrix}_{6\times 6} \begin{bmatrix} \mathbf{e} \\ \dot{\mathbf{e}} \end{bmatrix}_{6\times 1} + \begin{bmatrix} 0 \\ \left(\mathbf{e}^{T} - \mathbf{e}_{T}^{T} \right)^{\times} \mathbf{D} \mathbf{I}_{S}^{-1} \end{bmatrix}_{6\times 3} \begin{bmatrix} \mathbf{T}_{C} \end{bmatrix}_{3\times 1}$$

• For relative translational motion

$$\Delta \boldsymbol{\rho} = \boldsymbol{\rho}_{0} - \boldsymbol{\rho}_{d}$$

$$\begin{bmatrix} \Delta \dot{\boldsymbol{\rho}} \\ \Delta \ddot{\boldsymbol{\rho}} \end{bmatrix}_{6\times 1} = \begin{bmatrix} \mathbf{0} & \mathbf{E} \\ \mathbf{N} & \mathbf{M} \end{bmatrix}_{6\times 6} \begin{bmatrix} \Delta \boldsymbol{\rho} \\ \Delta \dot{\boldsymbol{\rho}} \end{bmatrix}_{6\times 1} + \begin{bmatrix} \mathbf{0} \\ \mathbf{E} \end{bmatrix}_{6\times 3} \begin{bmatrix} \mathbf{a} \end{bmatrix}_{3\times 1} - \dot{\boldsymbol{\omega}}_{T} \times \ddot{\boldsymbol{\rho}}_{d}^{T} + \boldsymbol{\omega}_{T} \times (\boldsymbol{\omega}_{T} \times \ddot{\boldsymbol{\rho}}_{d}^{T})$$



Attractive and repulsive virtual potentials control approach

Attractive and repulsive potential functions

$$V = V_a + V_r = -C_a e^{\frac{-r}{la}} + C_r e^{\frac{-r}{lr}}$$

- The force produced by the potential function
- $-\nabla V = F$
- **Dynamical equations** $\ddot{x} = n^2 x$

Attraction Repulsion Selective attraction

$$\begin{aligned}
& \text{tential functions} \\
& \overline{C}_r e^{\frac{-r}{lr}} \\
& \text{potential function} \\
& \overline{x} = n^2 x + 2n \dot{y} + \left(\frac{-C_a}{l_a} e^{\frac{-r}{l_a}} + \frac{C_r}{l_r} e^{\frac{r}{l_r}}\right) \frac{x}{r} + f_r e_r \frac{x}{|\mathbf{r}|} - f_r (V_{\tau x} - V_{\tau x d}), \\
& \overline{y} = -2n \dot{x} + \left(\frac{-C_a}{l_a} e^{\frac{-r}{l_a}} + \frac{C_r}{l_r} e^{\frac{r}{l_r}}\right) \frac{y}{r} + f_r e_r \frac{y}{|\mathbf{r}|} - f_r (V_{\tau y} - V_{\tau y d}), \\
& \overline{z} = -n^2 z + \left(\frac{-C_a}{l_a} e^{\frac{-r}{l_a}} + \frac{C_r}{l_r} e^{\frac{r}{l_r}}\right) \frac{z}{r} + f_r e_r \frac{z}{|\mathbf{r}|} - f_r (V_{\tau z} - V_{\tau x z}), \\
& 9
\end{aligned}$$

Numerical simulations examples

• Initial conditions and system parameters

h = 700 км; e = 0.03; $i = 70^{\circ}$ $\mathbf{I}_{T} = diag(1,1.4,1) \text{kg} \cdot \text{m}^{2}$ $\mathbf{I}_{s} = diag(2,4,4) \text{kg} \cdot \text{m}^{2}$ m = 45 kg $\mathbf{q}_{T0} = [0, 0, 1, 0]^T$ $\mathbf{q}_{s0} = [0.59, 0.2, 0.6, 0.5]^T$ $\omega_T^T = [0.2, 0.4, 0]^T \text{ rad/s}$ $\mathbf{\rho}_0 = r_0 = [x_0, y_0, z_0]^T = [5, 5, 5]^T \mathrm{m}$ $\dot{\mathbf{\rho}}_0 = \dot{r}_0 = [1, 2, 3]^T \,\mathrm{m/s}$ $\mathbf{\rho}_{i1} = [1, 1, 1]^T \mathrm{m}$ $\mathbf{\rho}_{io} = [1, 1, 1]^T \mathrm{m}$

• For virtual potentials

Attraction and repulsion

 $l_a, = 5, l_r = 2.755$ $C_a = 5, C_r = 4.5$ $f_r = 1,$

Selective attraction l_a , = 5, l_r = 2.755 C_a = 12, C_r = 1

 $C_a = 12, C_r$ $f_r = 0$

• For SDRE

Control parameters

 $Q_{attitude} = I_6,$ $R_{attitude} = 10^{-3} I_3,$ $Q_{translation} = 1I_6,$ $R_{translation} = 10I_3.$





Relative trajectory using SDRE-based control



Relative trajectory



Control force and torques: SDRE-based control

- The time of approaching is 14s
- The required control values are high
- No collision avoidance



Control force for translational motion

Control Torque for attitude motion



Trajectory and relative distance: virtual potentials





Control force and torque: virtual potentials

- The time of approaching is 1100s
- Low control values
- Collision avoidance included



Control force for translational motion

Control torque for rotational motion

Monte-Carlo study: tumbling debris angular velocity SDRE-based control



Required Delta-V vs the initial angular velocity of the debris

Approaching time vs the initial angular velocity of the debris

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Monte-Carlo study: tumbling debris angular velocity Virtual potentials control



Required Delta-V vs the initial angular velocity of the debris



Capturing time vs the initial angular velocity of the debris



Monte-Carlo study: maximum value of thrust

SDRE-based control



Delta-V vs the thrust value constraint

Approaching time vs the thrust value constraint



Monte-Carlo Study: maximum value of thrust virtual potentials control



Rendezvous Delta-V time vs the thrust value constraint



Approaching time vs the thrust value constraint



Conclusions

- Two types of control algorithms are developed: SDRE-based and virtual potentials approaches
- The algorithms performance in terms of required Delta-V and approaching time is compared
- The tumbling debris angular velocity greatly affects the cost of SDRE-control
- The virtual potential control takes long time to approach the debris

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